Lecture 35, Apr 8, 2022

Multivariable Optimization Example: Rocket Science

- Tsiolkovsky's Rocket Equation: $\Delta v = I_{sp} \ln \frac{m_i}{m_f}$
 - Δv is the momentum change given to the rocket
 - $-I_{sp}$ is the specific impulse of the fuel (i.e. speed of exhaust gases)
 - $-m_i$ and m_f are the initial and final mass
- Escape velocity of the Earth is about $\Delta v \approx 11$ km/s; $I_{sp} \approx 3.2$ km/s for common rocket fuel
- For a single stage rocket: $\frac{m_i}{m_f} = \exp\left(\frac{11}{3.2}\right) = 31.1 \implies \frac{m_{fuel}}{m_i} = \frac{m_i m_f}{m_i} = 1 \frac{m_f}{m_i} = 0.97$
 - -97% of the rocket would have to be fuel for the rocket to break escape velocity
 - Comparison: Aluminum beverage can is 96%
- What if we limit fuel to be 80% of the total weight?
- Break down $m_i = A + M_s + M_f$ where A is the payload, M_s is the rocket structure, M_{fuel} is the fuel - Define $S = \frac{M_s}{M_s + M_{fuel}}$ as the structure factor, i.e. the fraction of the mass without payload
 - that's the structure

- Define rocket mass
$$M_r = M_s + M_{fuel}$$

$$\frac{m_i}{m_f} = \frac{A + M_s + M_{fuel}}{A + M_s} = \frac{A + M_r}{A + SM_r} \implies \Delta v = I_{sp} \ln\left(\frac{A + M_r}{A + SM_r}\right)$$

* If we plug in S = 0.2 and I_{sp} from before, without any payload this would give $\Delta v = 5.15$ km/s • Given a 3-stage rocket, how can we optimize the sizes of the 3 stages to minimize the total rocket mass?

– Let the 3 stages have masses M_1, M_2, M_3 and the same structure factor S

- The first stage has initial mass $A + M_1 + M_2 + M_3$; after the fuel is expended, $M_1 \rightarrow SM_1$, then the second stage has mass $A + M_2 + M_3$ and so on
- $-\Delta v_{tot} = \Delta v_1 + \Delta v_2 + \Delta v_3$

$$= I_{sp} \left[\ln \left(\frac{A + M_1 + M_2 + M_3}{A + SM_1 + M_2 + M_3} \right) + \ln \left(\frac{A + M_2 + M_3}{A + SM_2 + M_3} \right) + \left(\frac{A + M_3}{A + SM_3} \right) \right]$$

• Problem: Given A, Δv , I_{sp} , and S for a 3 stage rocket, find M_1, M_2, M_3 such that $M_r =$ $f(M_1, M_2, M_3) = M_1 + M_2 + M_3$ is minimized, subject to the constraint of the rocket equation above

- Constraint:
$$g(M_1, M_2, M_3) = \ln\left(\frac{A + M_1 + M_2 + M_3}{A + SM_1 + M_2 + M_3}\right) + \ln\left(\frac{A + M_2 + M_3}{A + SM_2 + M_3}\right) + \left(\frac{A + M_3}{A + SM_3}\right) = \Delta v$$

$$\frac{\overline{I_{sp}}}{\nabla f} = (1, 1, 1)
- g = \ln(A + M_1 + M_2 + M_3) - \ln(A + SM_1 + M_2 + M_3) + \ln(A + M_2 + M_3) - \ln(A + SM_2 + M_3) + \ln(A + M_3) - \ln(A + SM_3)
- For $M_1 : \lambda \frac{\partial f}{\partial M_1} = \frac{\partial g}{\partial M_1} \implies \lambda = \frac{1}{A + M_1 + M_2 + M_3} - \frac{S}{A + SM_1 + M_2 + M_3} + \frac{1}{A + M_2 + M_3} - \frac{S}{A + SM_2 + M_3} + \frac{1}{A + M_2 + M_3} - \frac{S}{A + SM_2 + M_3} + \frac{1}{A + M_2 + M_3} - \frac{S}{A + SM_2 + M_3} + \frac{1}{A + M_2 + M_3} - \frac{1}{A + M_1 + M_2 + M_3} - \frac{1}{A + SM_1 + M_2 + M_3} + \frac{1}{A + M_2 + M_3} - \frac{1}{A + SM_1 + M_2 + M_3} + \frac{1}{A + M_2 + M_3} - \frac{1}{A + SM_1 + M_2 + M_3} + \frac{1}{A + M_2 + M_3} - \frac{S}{A + SM_2 + M_3} + \frac{1}{A + M_3} - \frac{S}{A + SM_3} = \frac{1}{A + M_1 + M_2 + M_3} - \frac{1}{A + SM_1 + M_2 + M_3} + \frac{1}{A + M_2 + M_3} - \frac{S}{A + SM_2 + M_3} + \frac{1}{A + M_3} - \frac{S}{A + SM_3} = \frac{1}{A + SM_2 + M_3} + \frac{1}{A + M_3} - \frac{S}{A + SM_3} = \frac{1}{A + SM_2 + M_3} + \frac{1}{A + M_3} - \frac{S}{A + SM_3} = \frac{1}{A + SM_2 + M_3} + \frac{1}{A + M_3} - \frac{S}{A + SM_3} = \frac{1}{A + SM_$$$

 $- \text{ Subtracting the third from the second equation gives: } 0 = \frac{1}{A + SM_2 + M_3} - \frac{S}{A + SM_2 + M_3} + \frac{S}{A + SM_3} - \frac{1}{A + M_3} \Longrightarrow -\frac{S}{A + SM_3} = \frac{1 - S}{A + SM_2 + M_3} - \frac{1}{A + M_3} + \frac{S(A + M_3)}{A + SM_3} = \frac{(1 - S)(A + M_3)}{A + SM_2 + M_3} - 1 = \frac{A + M_3 - SA - SM_3 - A - SM_2 - M_3}{A + SM_2 + M_3} =$

$$-\frac{S(A+M_2+M_3)}{A+SM_2+M_3} \\ * \frac{A+M_3}{A+SM_3} = \frac{A+M_2+M_3}{A+SM_2+M_3}$$

* $\frac{A + M_3}{A + SM_3} = \frac{A + M_2 + M_3}{A + SM_2 + M_3}$ * Note these are the things inside ln in the second and third terms in g

- Subtracting the second equation from the first and going through the same steps, we obtain the relation $\frac{A + M_2 + M_3}{A + SM_2 + M_3} + \frac{A + M_1 + M_2 + M_3}{A + SM_1 + M_2 + M_3}$ * These are the arguments in ln in the first and second terms in g - This basically tells us that $\frac{m_i}{m_f}$ should be the same for each stage of the rocket

$$- \text{Let } N = \frac{A + M_3}{A + SM_3} \implies \frac{\Delta v}{I_{sp}} = 3 \ln N \implies N = \exp\left(\frac{\Delta v}{3I_{sp}}\right) \\ - M_3 = A \frac{N - 1}{1 - SN}, M_2 = (A + M_3) \frac{N - 1}{1 - SN}, M_1 = (A + M_2 + M_3) \frac{N - 1}{1 - SN}$$